

Large Eddy Simulation of advanced injection strategies under liquid rocket engine operating conditions

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In this contribution, we present numerical simulations of advanced injection strategies for liquid rocket engines (LRE), featuring supercritical pressure conditions and oxygen-methane as a propellant combination. In particular, we investigate: (i) the crossflow injection of cryogenic propellants under transcritical conditions, and (ii) a generic tricoaxial injection strategy for methane and oxygen as propellants. These two paradigmatic configurations are chosen as representatives of advanced injection strategies and, therefore, investigated using medium to high fidelity numerical simulations.

The crossflow injection has been widely investigated under standard thermodynamic conditions in the context of ideal gases, while conditions of relevance for LRE are significantly less investigated [1]. This promotes a significant interest in simulating such a configuration under LRE conditions and developing new surrogate models for jet development and penetration. Regarding the second configuration discussed, recent experiments have shown that the tri-coaxial injection enhances the combustion performance under LRE conditions by means of a rapid near-field mixing [2], motivating a substantial technological interest. Starting from these observations, numerical investigations of the two configurations are carried out in the present contribution. The simulations are performed using Large Eddy Simulations (LES) and a consistent flamelet-based turbulent combustion model implemented in the OpenFOAM framework and developed in the context of LRE injection [3] and heat transfer [4].

References

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